

Minimizing Oscillations for Magnitude and Rate-Saturated Discrete-Time Systems by a D_R Region Pole Placement

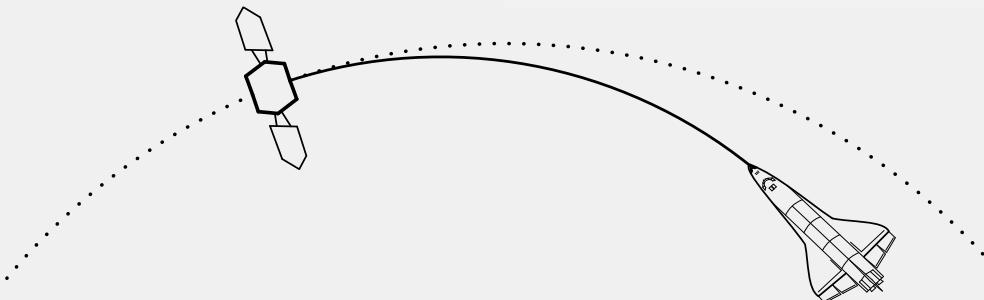
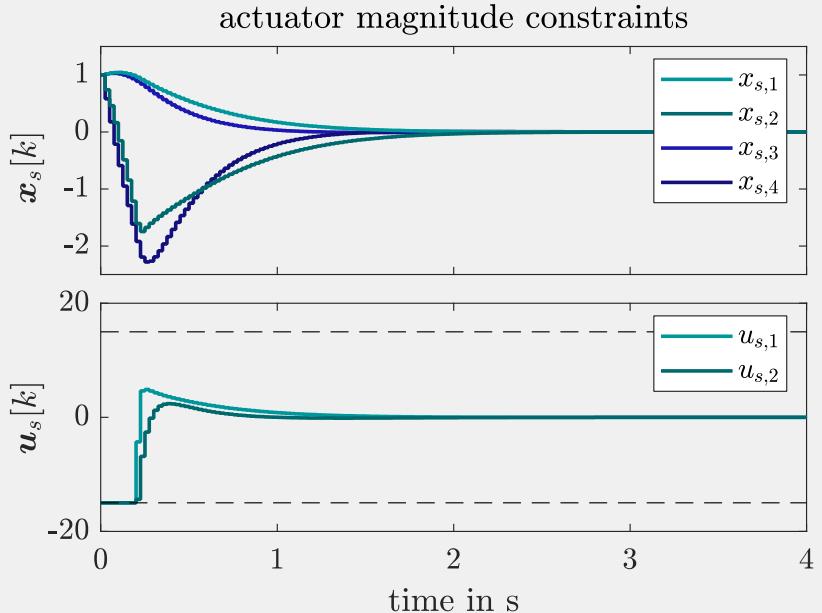
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Sabine Lerch

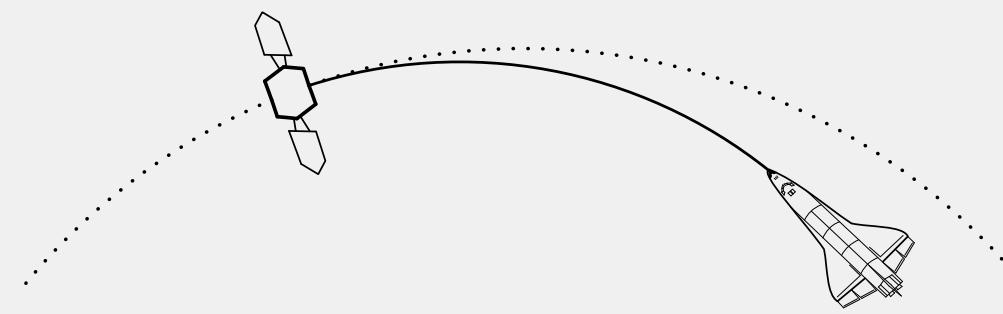
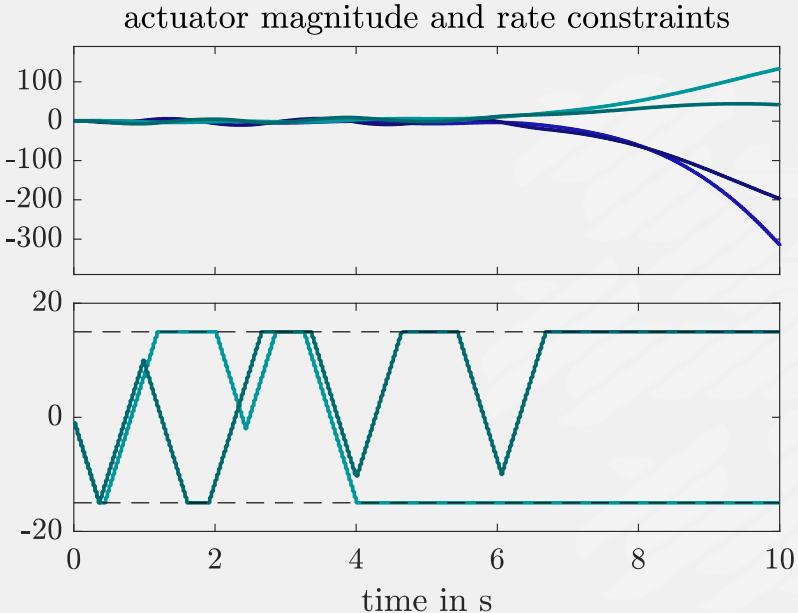
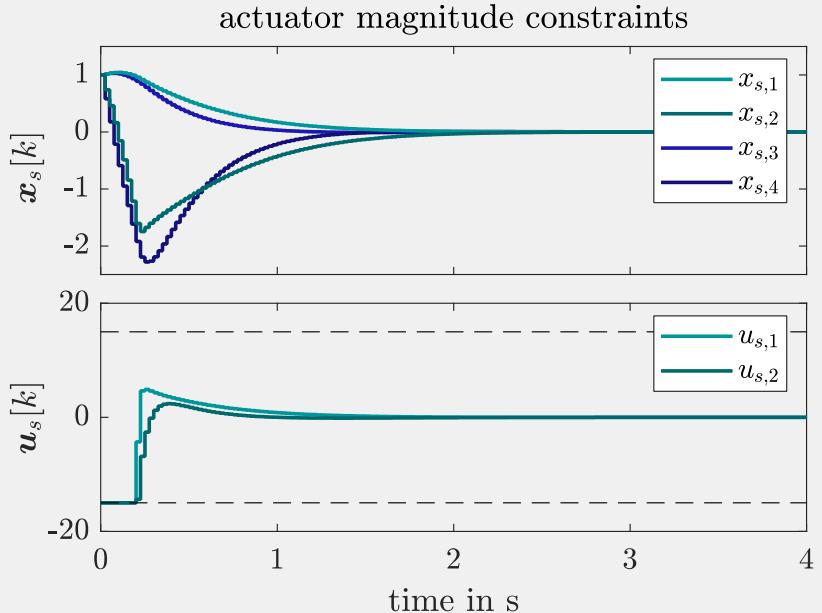
*Chair of Automation Engineering and Control Systems
School of Electrical, Information and Media Engineering
Bergische Universität Wuppertal, Germany*

in collaboration with Robert Dehnert, Michelle Rosik and Bernd Tibken

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Motivation: Stability of Systems with Actuator Constraints

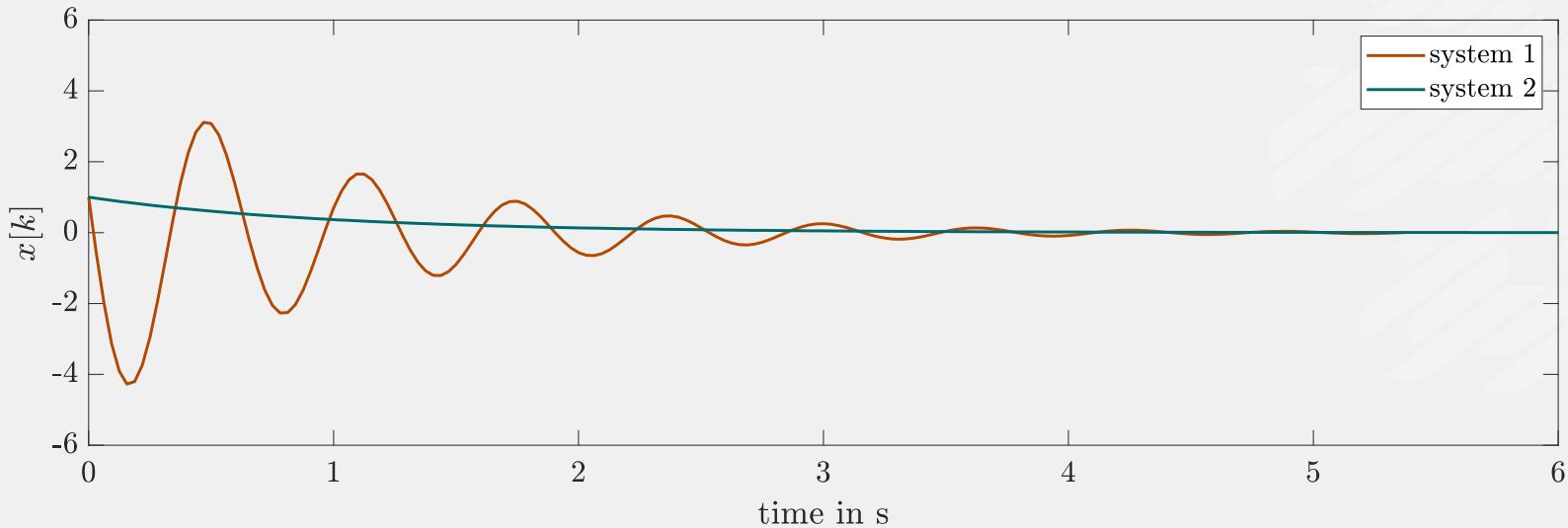


Decay rate: maximum value of σ such that

$$\|x[k]\| \leq me^{-\sigma k} \|x[0]\|$$

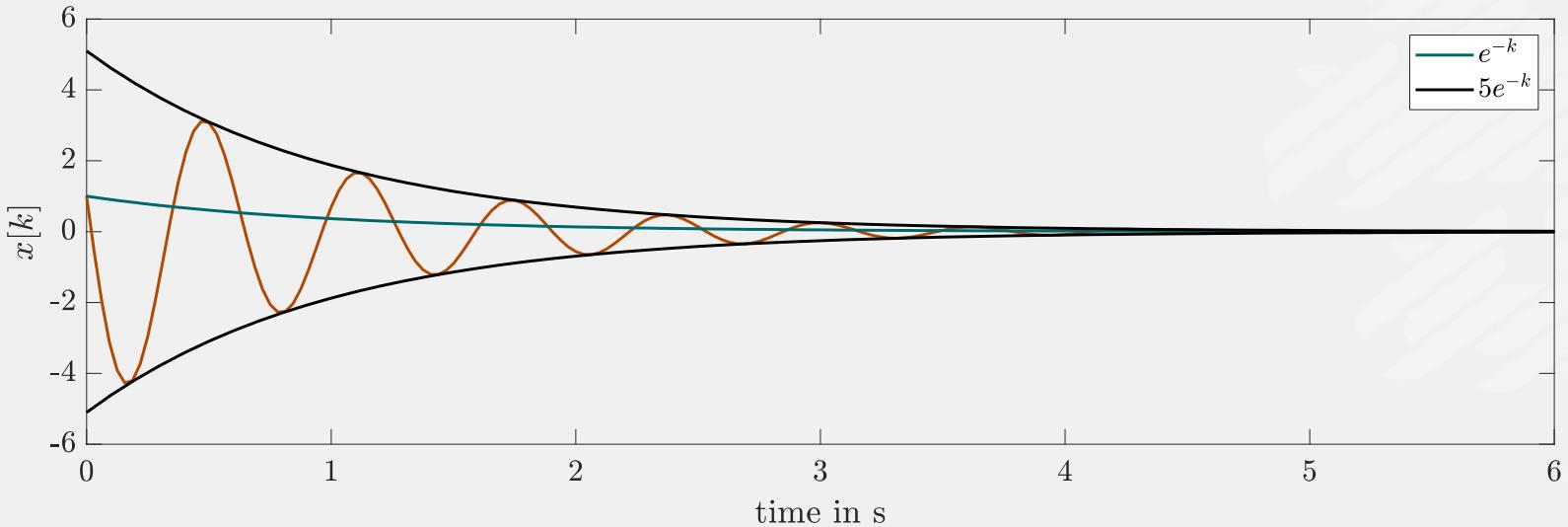
Decay rate: maximum value of σ such that

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Decay rate: maximum value of σ such that

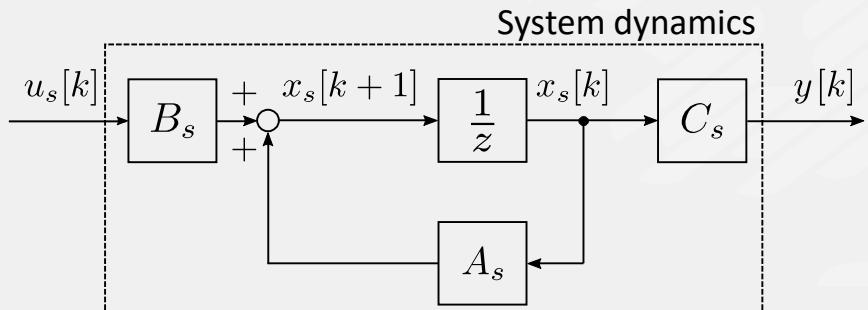
$$\|x[k]\| \leq m e^{-\sigma k} \|x[0]\|$$



1. System description
 - Linear discrete time system
 - Nonlinear actuator with magnitude and rate saturation (MRS)
 - Different controller types
2. Stability conditions for MRS Systems
3. Maximization of the decay rate
4. Maximization of the damping
5. Iterative LMI Algorithm
6. Examples
7. Conclusion

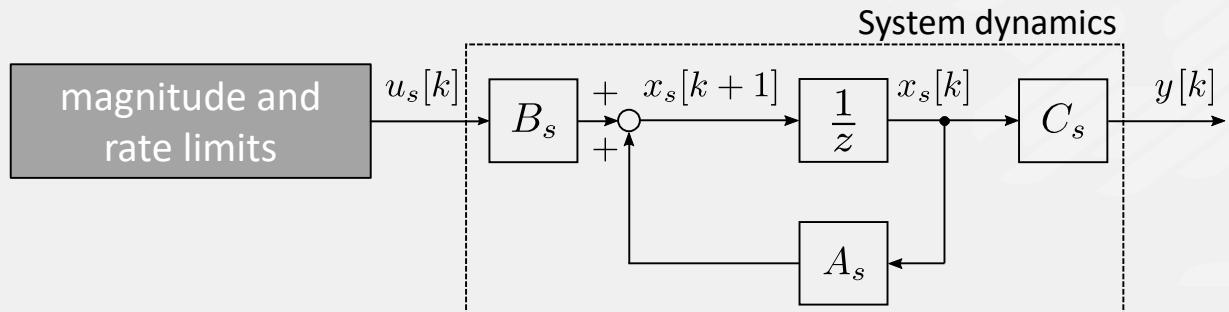
$$x_s[k+1] = A_s x_s[k] + B_s u_s[k] \text{ with } x_s[0] = x_{s,0}$$

$$y[k] = C_s x_s[k]$$



$$x_s[k+1] = A_s x_s[k] + B_s u_s[k] \text{ with } x_s[0] = x_{s,0}$$

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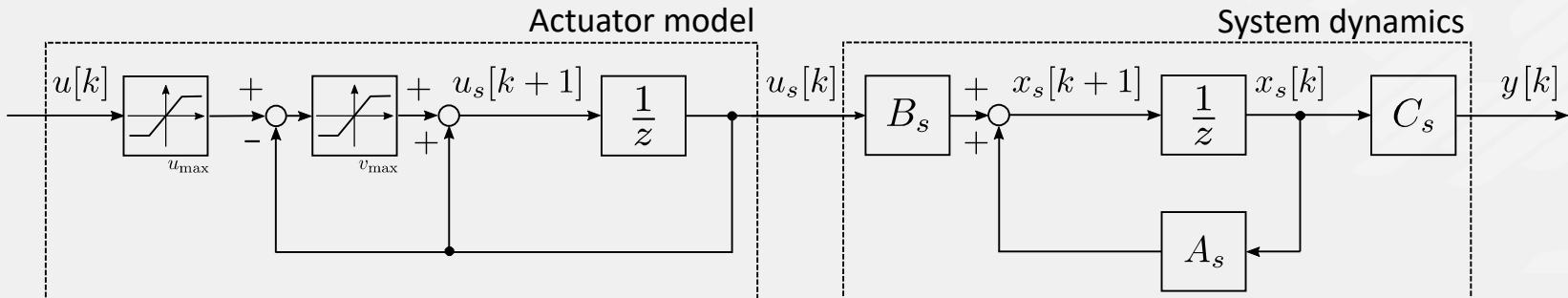


$$x_s[k+1] = A_s x_s[k] + B_s u_s[k] \text{ with } x_s[0] = x_{s,0}$$

$$y[k] = C_s x_s[k]$$

$$u_s[k+1] = u_s[k] + \text{sat}_V(\text{sat}_U(u[k]) - u_s[k])$$

U : magnitude saturation
 V : rate saturation



$$x_s[k+1] = A_s x_s[k] + B_s u_s[k]$$

$$u_s[k+1] = u_s[k] + \text{sat}_V(\text{sat}_U(u[k]) - u_s[k])$$

$$y[k] = C_s x_s[k]$$

Augmented state vector

$$x[k] = \begin{pmatrix} x_s[k] \\ u_s[k] \end{pmatrix}$$

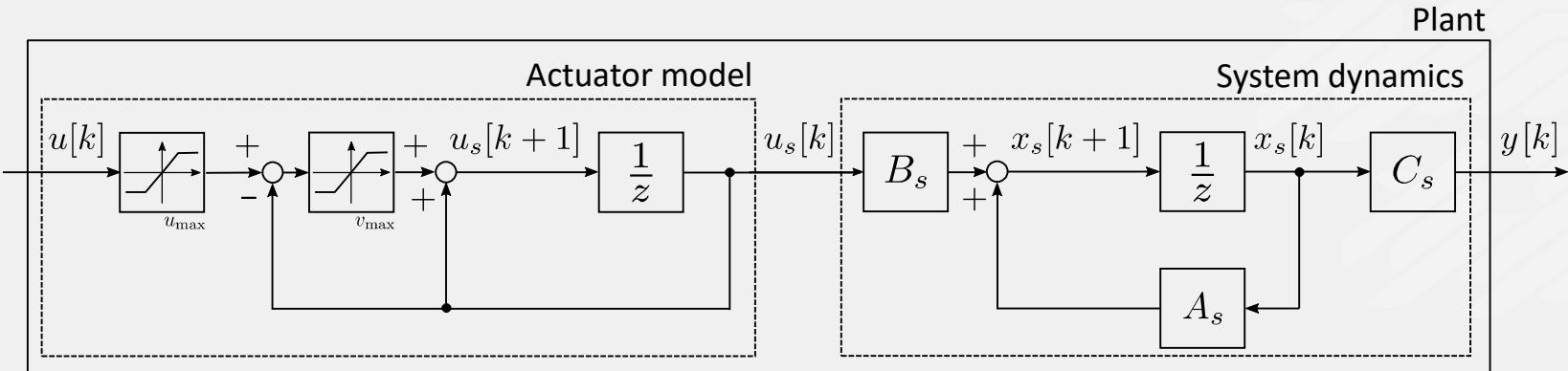
Augmented system

$$\boxed{x[k+1] = Ax[k] + B\text{sat}_V(\text{sat}_U(u[k]) + Fx[k]) \\ y[k] = Cx[k]}$$

$$A = \begin{pmatrix} A_s & B_s \\ 0 & I \end{pmatrix}, B = \begin{pmatrix} 0 \\ I \end{pmatrix}, F = (0 \quad -I), C = (C_s \quad 0)$$

$$x[k+1] = Ax[k] + B\text{sat}_V(\text{sat}_U(u[k]) + Fx[k])$$

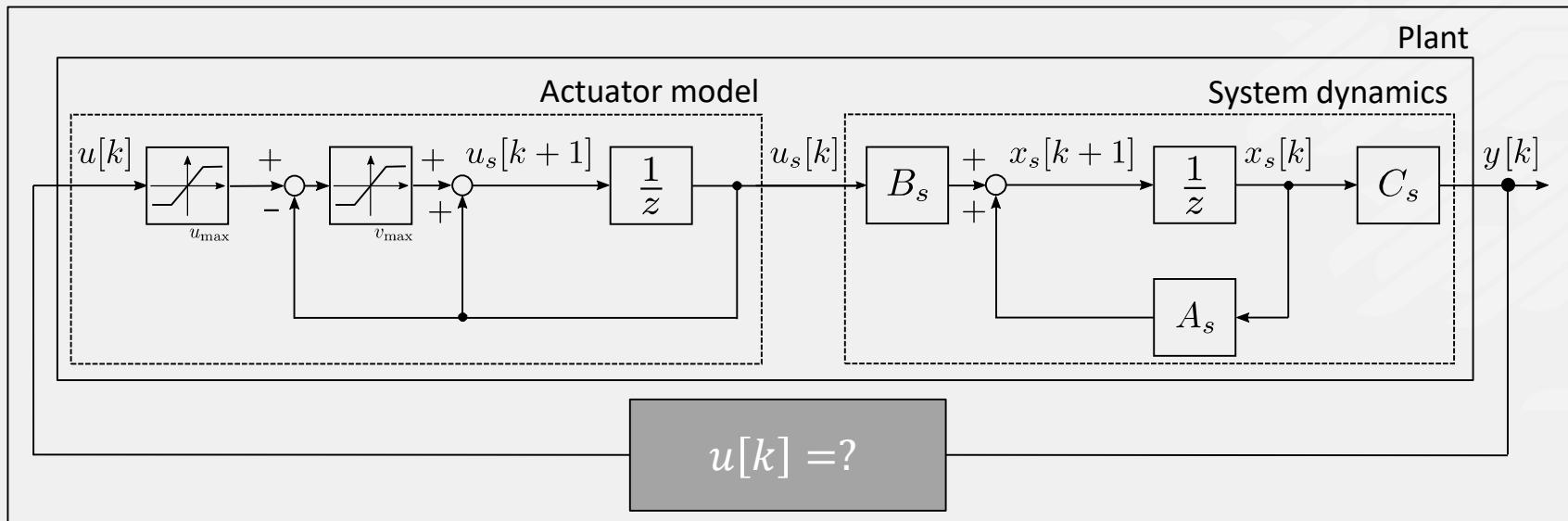
$$y[k] = Cx[k]$$



$$x[k+1] = Ax[k] + B\text{sat}_V(\text{sat}_U(u[k]) + Fx[k])$$

$$y[k] = Cx[k]$$

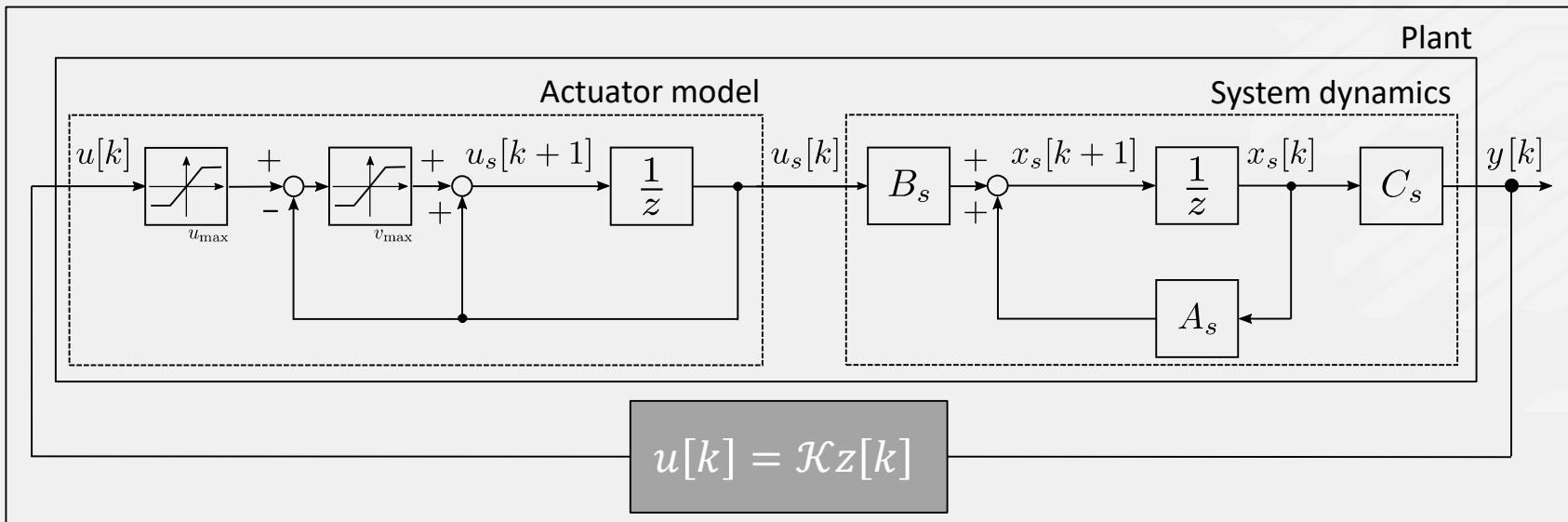
Closed Loop



$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

$$z[k] = \begin{pmatrix} x_s[k] \\ u_s[k] \\ x_c[k] \end{pmatrix}$$

Closed Loop



$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Controller type	\mathcal{A}	\mathcal{B}	\mathcal{K}	\mathcal{F}
Full state feedback	A	B	K	$(0 \quad -I)$
Static output feedback	A	B	KC	$(0 \quad -I)$
Observer-based feedback	$\begin{pmatrix} A & 0 \\ 0 & A_s - LC_s \end{pmatrix}$	$\begin{pmatrix} B \\ 0 \end{pmatrix}$	$(K \quad 0 \quad -K)$	$(0 \quad -I)$
Dynamic output feedback	$\begin{pmatrix} A & 0 \\ B_c C & A_c \end{pmatrix}$	$\begin{pmatrix} B \\ 0 \end{pmatrix}$	$(D_c C \quad C_c)$	$(0 \quad -I)$
PID control	$\begin{pmatrix} A & 0 & 0 \\ -C & I & 0 \\ -C & 0 & 0 \end{pmatrix}$	$\begin{pmatrix} B \\ 0 \\ 0 \end{pmatrix}$	$((K_P + K_D)C \quad K_I \quad -K_D)$	$(0 \quad -I)$
:	:	:	:	:

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Controller type	\mathcal{A}	\mathcal{B}	\mathcal{K}	\mathcal{F}
Full state feedback	A	B	K	$(0 \quad -I)$
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PID control	$\begin{pmatrix} A & 0 & 0 \\ -C & I & 0 \\ -C & 0 & 0 \end{pmatrix}$	$\begin{pmatrix} B \\ 0 \\ 0 \end{pmatrix}$	$((K_P + K_D)C \quad K_I \quad -K_D)$	$(0 \quad -I)$
:	:	:	:	:

Quadratic Lyapunov Function

$$V(z[k]) = z^T[k] P z[k]$$

Domain of Attraction

$$\mathcal{E}(P) = \{z \in \mathbb{R}^{n_z} : z^T P z \leq 1\}$$

Stability Condition

$$z^T[k+1] P z[k+1] - z^T[k] P z[k] < 0$$

Linear autonomous system

$$z[k+1] = \mathcal{A}z[k]$$

Stability Condition

$$\mathcal{A}^T P \mathcal{A} - P < 0$$

Quadratic Lyapunov Function

$$V(z[k]) = z^T[k] P z[k]$$

Domain of Attraction

$$\mathcal{E}(P) = \{z \in \mathbb{R}^{n_z} : z^T P z \leq 1\}$$

Stability Condition

$$z^T[k+1] P z[k+1] - z^T[k] P z[k] < 0$$

Nonlinear system

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Stability Condition

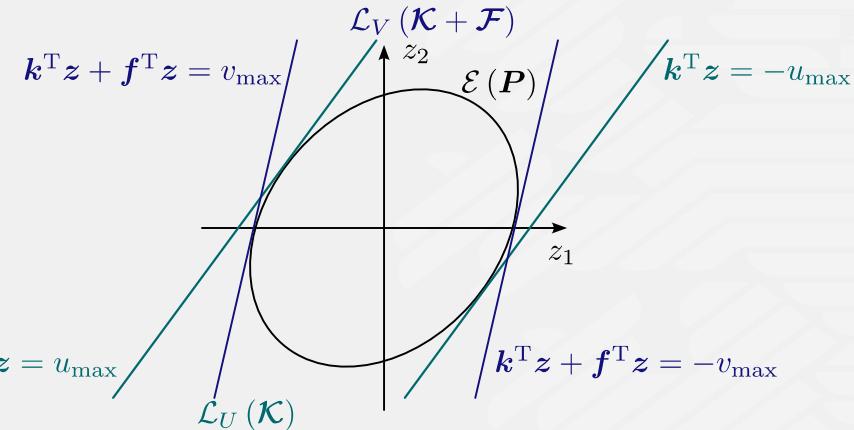
$$\begin{aligned} & (\mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]))^T P (\mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])) \\ & - z^T[k] P z[k] < 0 \end{aligned}$$

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Linear when not saturated

$$\text{sat}_U(\mathcal{K}z[k]) = \mathcal{K}z[k]$$

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) = (\mathcal{K} + \mathcal{F})z[k]$$



$$\mathcal{L}_U(\mathcal{K}) = \{z \in \mathbb{R}^{n_z} \mid |k_q^T z| < u_{\max,q}, q = 1, \dots, m\}$$

$$\mathcal{L}_V(\mathcal{K} + \mathcal{F}) = \{z \in \mathbb{R}^{n_z} \mid |k_q^T z + f_q^T z| < v_{\max,q}, q = 1, \dots, m\}$$

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Linear when not saturated

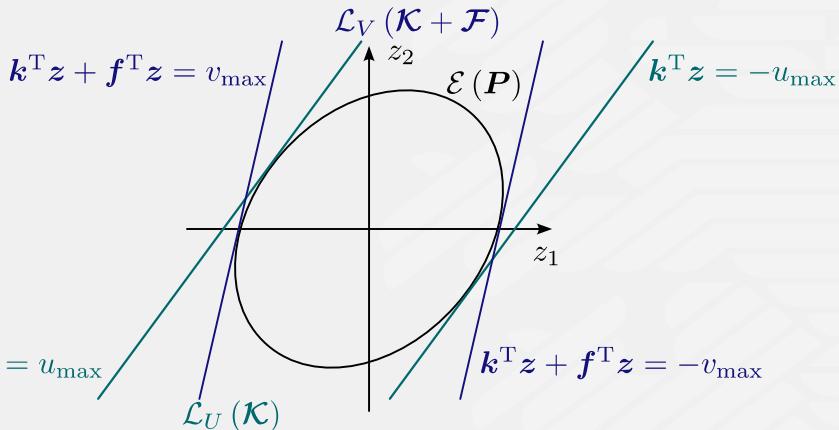
$$\text{sat}_U(\mathcal{K}z[k]) = \mathcal{K}z[k]$$

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) = (\mathcal{K} + \mathcal{F})z[k]$$

Stability conditions

$$(\mathcal{A} + \mathcal{B}(\mathcal{K} + \mathcal{F}))^\top P(\mathcal{A} + \mathcal{B}(\mathcal{K} + \mathcal{F})) - P < 0$$

$$\mathcal{E}(P) \subseteq \mathcal{L}_U(\mathcal{K}) \cap \mathcal{L}_V(\mathcal{K} + \mathcal{F})$$



$$\mathcal{L}_U(\mathcal{K}) = \{z \in \mathbb{R}^{n_z} \mid |k_q^T z| < u_{\max,q}, q = 1, \dots, m\}$$

$$\mathcal{L}_V(\mathcal{K} + \mathcal{F}) = \{z \in \mathbb{R}^{n_z} \mid |k_q^T z + f_q^T z| < v_{\max,q}, q = 1, \dots, m\}$$

Saturated controller

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Auxiliary controllers

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{H}_1 z[k]) + \mathcal{F}z[k])$$

$$u_s[k] = \text{sat}_V(\mathcal{H}_2 z[k])$$

with $\text{sat}_U(\mathcal{H}_1 z[k]) = \mathcal{H}_1 z[k]$

$$\text{sat}_V(\mathcal{H}_2 z[k]) = \mathcal{H}_2 z[k]$$

Saturated controller

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Auxiliary controllers

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{H}_1 z[k]) + \mathcal{F}z[k])$$

$$u_s[k] = \text{sat}_V(\mathcal{H}_2 z[k])$$

with $\text{sat}_U(\mathcal{H}_1 z[k]) = \mathcal{H}_1 z[k]$

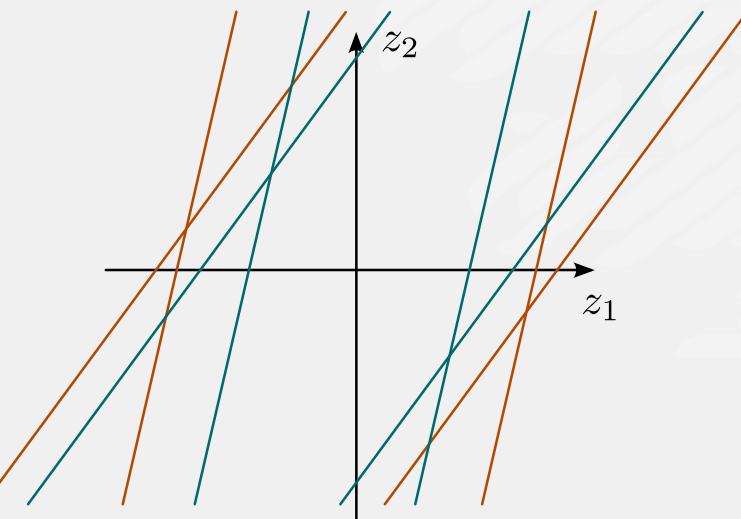
$\text{sat}_V(\mathcal{H}_2 z[k]) = \mathcal{H}_2 z[k]$

$$\mathcal{L}_U(\mathcal{K}) = \{z \in \mathbb{R}^{n_z} \mid |\mathcal{K}_q^T z| < u_{\max,q}, q = 1, \dots, m\}$$

$$\mathcal{L}_V(\mathcal{K} + \mathcal{F}) = \{z \in \mathbb{R}^{n_z} \mid |\mathcal{K}_q^T z + \mathcal{F}_q^T z| < v_{\max,q}, q = 1, \dots, m\}$$

$$\mathcal{L}_U(\mathcal{H}_1) = \{z \in \mathbb{R}^{n_z} \mid |\mathcal{H}_{1,q}^T z| < u_{\max,q}, q = 1, \dots, m\}$$

$$\mathcal{L}_V(\mathcal{H}_2) = \{z \in \mathbb{R}^{n_z} \mid |\mathcal{H}_{2,q}^T z| < v_{\max,q}, q = 1, \dots, m\}$$



Saturated controller

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Auxiliary controllers

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{H}_1 z[k]) + \mathcal{F}z[k])$$

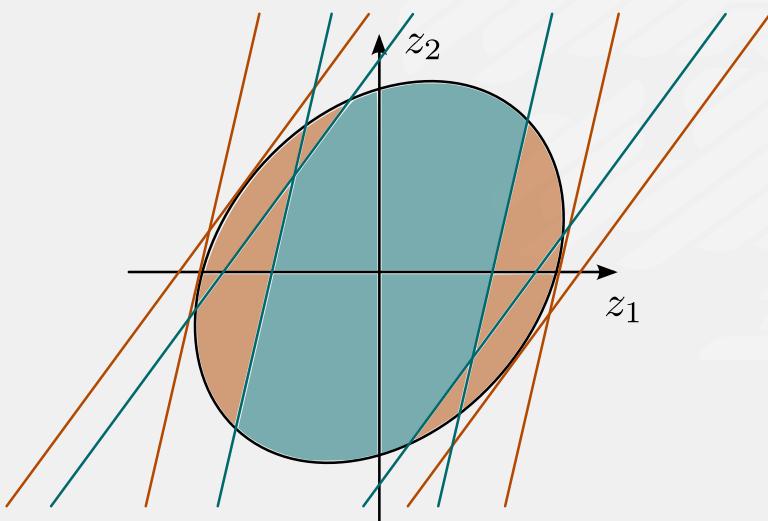
$$u_s[k] = \text{sat}_V(\mathcal{H}_2 z[k])$$

with $\text{sat}_U(\mathcal{H}_1 z[k]) = \mathcal{H}_1 z[k]$

$\text{sat}_V(\mathcal{H}_2 z[k]) = \mathcal{H}_2 z[k]$

$$\Rightarrow \mathcal{E}(P) \subseteq \mathcal{L}_V(\mathcal{H}_1) \cap \mathcal{L}_U(\mathcal{H}_2)$$

$$\begin{aligned}\mathcal{L}_U(\mathcal{K}) &= \{z \in \mathbb{R}^{n_z} \mid |\mathcal{K}^T z| < u_{\max,q}, q = 1, \dots, m\} \\ \mathcal{L}_V(\mathcal{K} + \mathcal{F}) &= \{z \in \mathbb{R}^{n_z} \mid |\mathcal{K}^T z + \mathcal{F}^T z| < v_{\max,q}, q = 1, \dots, m\} \\ \mathcal{L}_U(\mathcal{H}_1) &= \{z \in \mathbb{R}^{n_z} \mid |\mathcal{H}_{1,q}^T z| < u_{\max,q}, q = 1, \dots, m\} \\ \mathcal{L}_V(\mathcal{H}_2) &= \{z \in \mathbb{R}^{n_z} \mid |\mathcal{H}_{2,q}^T z| < v_{\max,q}, q = 1, \dots, m\}\end{aligned}$$



Saturated controller

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

Linear case

$$u_s[k] = (\mathcal{K} + \mathcal{F})z[k]$$

Auxiliary controllers

$$u_s[k] = \text{sat}_V(\text{sat}_U(\mathcal{H}_1 z[k]) + \mathcal{F}z[k])$$

$$u_s[k] = (\mathcal{H}_1 + \mathcal{F})z[k]$$

$$u_s[k] = \text{sat}_V(\mathcal{H}_2 z[k])$$

$$u_s[k] = \mathcal{H}_2 z[k]$$

with $\text{sat}_U(\mathcal{H}_1 z[k]) = \mathcal{H}_1 z[k]$

$\text{sat}_V(\mathcal{H}_2 z[k]) = \mathcal{H}_2 z[k]$

Convex combination of linear feedbacks

$$\Xi_i = D_{i,1}(\mathcal{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2, i = 1, \dots, 3^m$$

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) \in \text{co}\{\Xi_i z[k], i = 1, \dots, 3^m\}$$

Convex combination of linear feedbacks

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) \in \text{co}\{\Xi_i z[k], i = 1, \dots, 3^m\}$$

$$\text{with } \Xi_i = D_{i,1}(\mathcal{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2, i = 1, \dots, 3^m$$

Nonlinear system

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

$$\epsilon \text{ co}\{\tilde{\mathcal{A}}_i z[k] = \mathcal{A}z[k] + \mathcal{B}\Xi_i z[k], i = 1, \dots, 3^m\}$$

Convex combination of linear feedbacks

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) \in \text{co}\{\Xi_i z[k], i = 1, \dots, 3^m\}$$

$$\text{with } \Xi_i = D_{i,1}(\mathcal{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2, i = 1, \dots, 3^m$$

Nonlinear system

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

$$\epsilon \text{ co}\{\tilde{\mathcal{A}}_i z[k] = \mathcal{A}z[k] + \mathcal{B}\Xi_i z[k], i = 1, \dots, 3^m\}$$

Stability conditions

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m$$

$$\mathcal{E}(P) \subseteq \mathcal{L}_V(\mathcal{H}_1) \cap \mathcal{L}_U(\mathcal{H}_2)$$

Convex combination of linear feedbacks

$$\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k]) \in \text{co}\{\mathcal{E}_i z[k], i = 1, \dots, 3^m\}$$

$$\text{with } \mathcal{E}_i = D_{i,1}(\mathcal{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2, i = 1, \dots, 3^m$$

Nonlinear system

$$z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathcal{K}z[k]) + \mathcal{F}z[k])$$

$$\epsilon \text{ co}\{\tilde{\mathcal{A}}_i z[k] = \mathcal{A}z[k] + \mathcal{B}\mathcal{E}_i z[k], i = 1, \dots, 3^m\}$$

Stability conditions

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m$$

$$\mathcal{E}(P) \subseteq \mathcal{L}_V(\mathcal{H}_1) \cap \mathcal{L}_U(\mathcal{H}_2)$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Full state feedback: $\mathbf{z}[k+1] = \mathcal{A}\mathbf{z}[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathbf{K}\mathbf{z}[k]) + \mathcal{F}\mathbf{z}[k])$

$$\begin{pmatrix} P^{-1} & \mathcal{A} + \mathcal{B}(D_{i,1}(\mathbf{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2) \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

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$$\begin{pmatrix} P^{-1} & \mathcal{A} + \mathcal{B}(D_{i,1}(\mathbf{K} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2) \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Standard approach (e.g. Bateman and Lin (2002))

Change of variables $Q = P^{-1}, Y = KQ, G_1 = \mathcal{H}_1 Q, G_2 = \mathcal{H}_2 Q$

$$\Rightarrow \begin{pmatrix} Q & \mathcal{A}Q + \mathcal{B}(D_{i,1}(Y + \mathcal{F}Q) + D_{i,2}(G_1 + \mathcal{F}Q) + D_{i,3}G_2) \\ * & Q \end{pmatrix} > 0, i = 1, \dots, 3^m$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Static output feedback: $z[k+1] = \mathcal{A}z[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathbf{K}Cz[k]) + \mathcal{F}z[k])$

$$\begin{pmatrix} P^{-1} & \mathcal{A} + \mathcal{B}(D_{i,1}(\mathbf{K}C + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2) \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Bateman and Lin (2002): $Y = KCQ$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Static output feedback: $\mathbf{z}[k+1] = \mathcal{A}\mathbf{z}[k] + \mathcal{B}\text{sat}_V(\text{sat}_U(\mathbf{K}\mathbf{C}\mathbf{z}[k]) + \mathcal{F}\mathbf{z}[k])$

$$\begin{pmatrix} P^{-1} & \mathcal{A} + \mathcal{B}(D_{i,1}(\mathbf{K}\mathbf{C} + \mathcal{F}) + D_{i,2}(\mathcal{H}_1 + \mathcal{F}) + D_{i,3}\mathcal{H}_2) \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Standard approach (Crusius and Trofino (1999))

Change of variables $Q = P^{-1}, Y = KN, G_1 = \mathcal{H}_1 Q, G_2 = \mathcal{H}_2 Q$

$$\Rightarrow \begin{pmatrix} Q & \mathcal{A}Q + \mathcal{B}(D_{i,1}(YC + \mathcal{F}Q) + D_{i,2}(G_1 + \mathcal{F}Q) + D_{i,3}G_2) \\ * & Q \end{pmatrix} > 0, i = 1, \dots, 3^m$$

$$\mathbf{N}\mathbf{C} - \mathbf{C}Q = 0$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Linearization of P^{-1} (Dehnert (2020))

$$L = \hat{P}^{-1}(2I - P\hat{P}^{-1}) \leq P^{-1} \quad \left(\begin{matrix} L & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0 \Rightarrow \left(\begin{matrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0$$

$$\left(\begin{matrix} \hat{P}^{-1}(2I - P\hat{P}^{-1}) & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0, \quad i = 1, \dots, 3^m$$

with $\hat{P} = \hat{P}^T = \text{const}$

$$\hat{P} = P$$

$$\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i - P < 0, i = 1, \dots, 3^m \quad \Rightarrow \begin{pmatrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{pmatrix} > 0, i = 1, \dots, 3^m$$

Linearization of P^{-1} (Dehnert (2020))

$$L = \hat{P}^{-1}(2I - P\hat{P}^{-1}) \leq P^{-1} \quad \left(\begin{matrix} L & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0 \Rightarrow \left(\begin{matrix} P^{-1} & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0$$

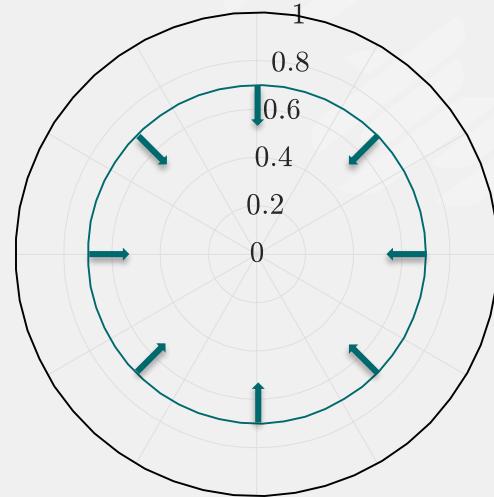
$$\left(\begin{matrix} \hat{P}^{-1}(2I - P\hat{P}^{-1}) & \tilde{\mathcal{A}}_i \\ * & P \end{matrix} \right) > 0, \quad i = 1, \dots, 3^m$$

Maximization of the decay rate

$$\min \bar{r}$$

s.t. $\left(\begin{matrix} \hat{P}^{-1}(2I - P\hat{P}^{-1}) & \tilde{\mathcal{A}}_i \\ * & \bar{r}^2 P \end{matrix} \right) > 0, \quad i = 1, \dots, 3^m$

with $\hat{P} = \hat{P}^T = \text{const}$



Linear auxiliary controllers

$$\mathcal{E}(P) \subseteq \mathcal{L}_V(\mathcal{H}_1) \cap \mathcal{L}_U(\mathcal{H}_2)$$

$$\begin{pmatrix} W_1 & \mathcal{H}_1 \\ \star & P \end{pmatrix} \succ 0,$$
$$\begin{pmatrix} W_2 & \mathcal{H}_2 \\ \star & P \end{pmatrix} \succ 0,$$

$$w_{1,qq} < u_{\max,q}^2, \quad q = 1, \dots, m,$$

$$w_{2,qq} < v_{\max,q}^2, \quad q = 1, \dots, m$$

Linear auxiliary controllers

$$\mathcal{E}(P) \subseteq \mathcal{L}_V(\mathcal{H}_1) \cap \mathcal{L}_U(\mathcal{H}_2)$$

$$\begin{pmatrix} W_1 & \mathcal{H}_1 \\ \star & P \end{pmatrix} \succ 0,$$

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$$w_{1,qq} < u_{\max,q}^2, \quad q = 1, \dots, m,$$

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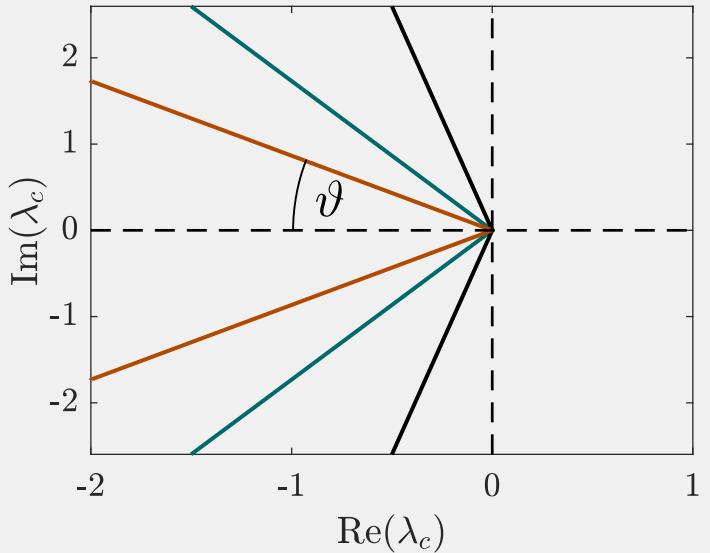
Initial state area

$$\mathcal{X}_0 = \text{co} \left\{ x_{0,1}, \dots, x_{0,N_{x_0}} \right\}$$

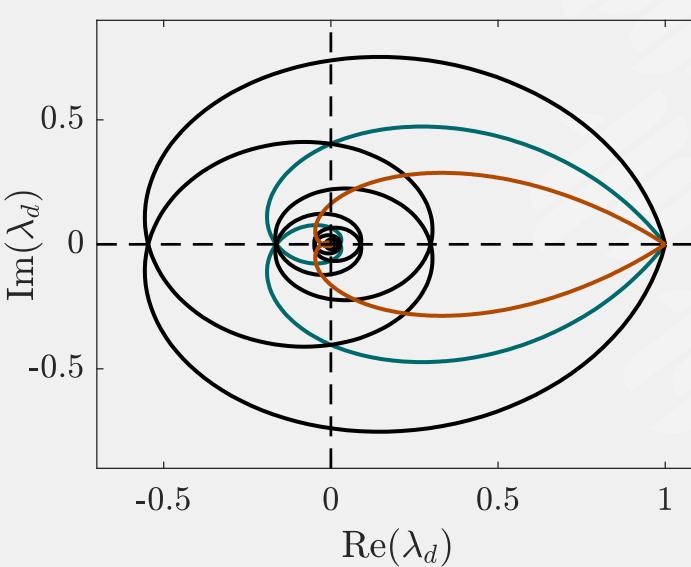
$$\mathcal{X}_0 \subseteq \mathcal{E}(P)$$

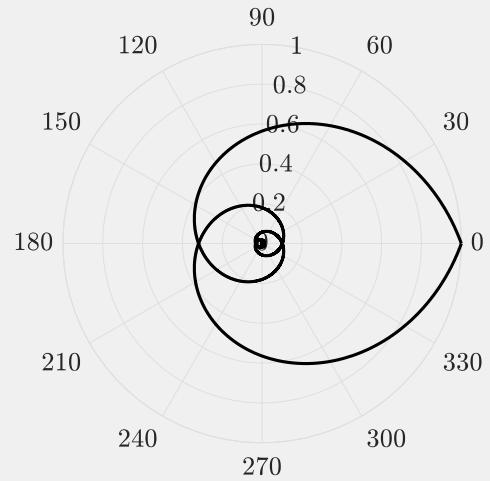
$$\begin{pmatrix} 1 & x_{0,s}^T P \\ \star & P \end{pmatrix} \succ 0, \quad s = 1, \dots, N_{x_0}$$

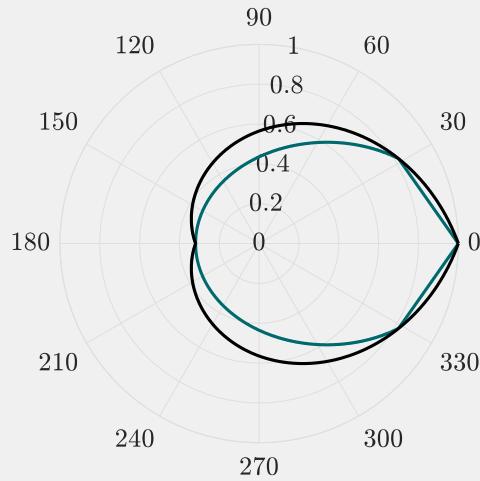
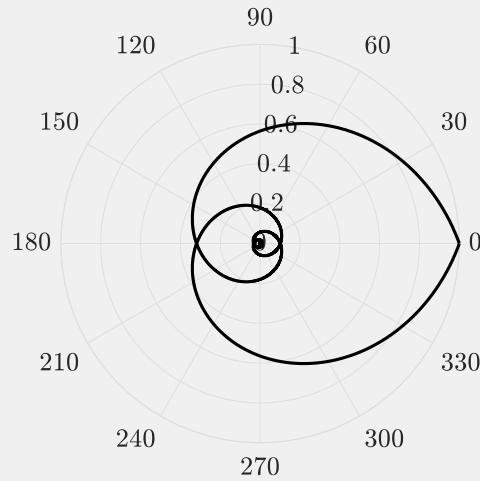
Continuous



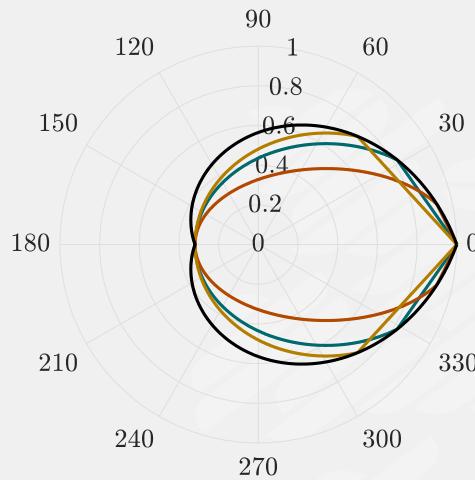
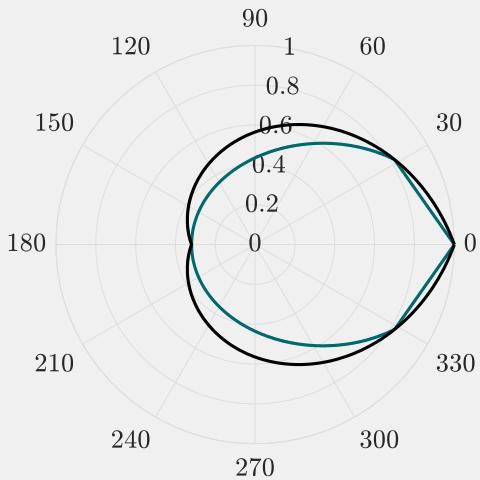
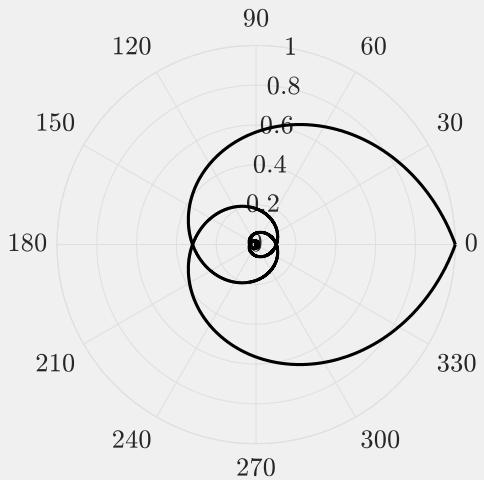
Discrete







Angle-Ellipse

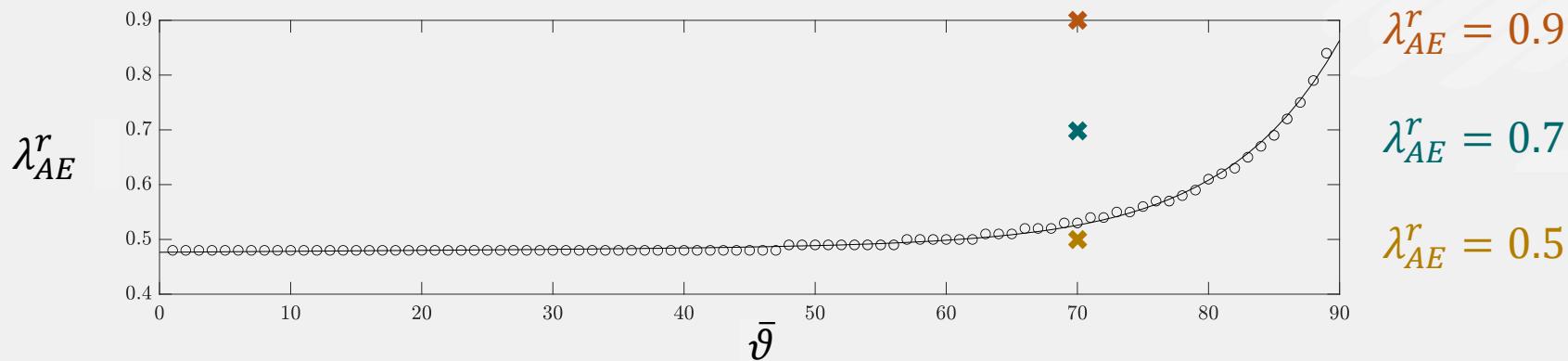
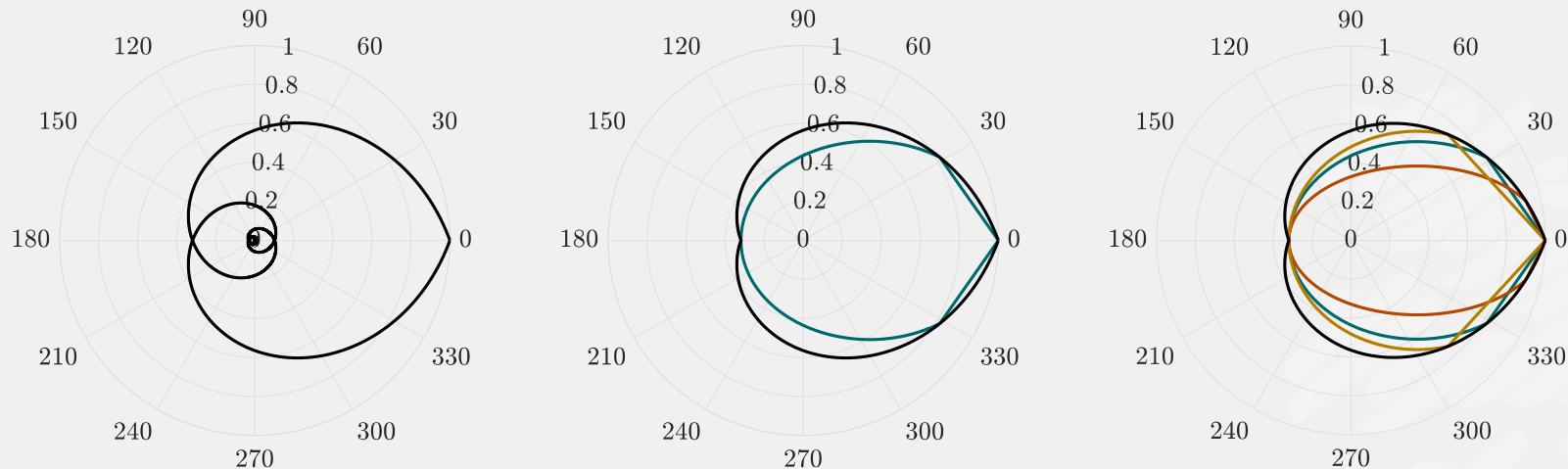


$$\lambda_{AE}^r = 0.9$$

$$\lambda_{AE}^r = 0.7$$

$$\lambda_{AE}^r = 0.5$$

Angle-Ellipse



$$D_R = \{z \in \mathbb{C} : R_{11} + R_{12}z + R_{12}^T z^* + R_{22}zz^* < 0\}$$

$$z = \operatorname{Re}(z) + j \operatorname{Im}(z), \quad z^* = \operatorname{Re}(z) - j \operatorname{Im}(z)$$

$$D_R = \{z \in \mathbb{C} : R_{11} + R_{12}z + R_{12}^T z^* + R_{22}zz^* < 0\}$$

$$z = \operatorname{Re}(z) + j \operatorname{Im}(z), \quad z^* = \operatorname{Re}(z) - j \operatorname{Im}(z)$$

Circle with upper bound
of spectral radii \bar{r}

$$R_{11} = -\bar{r}^2$$

$$R_{12} = 0$$

$$R_{22} = 1$$

$$\implies \operatorname{Re}^2 + \operatorname{Im}^2 < \bar{r}^2$$

$$D_R = \{z \in \mathbb{C} : R_{11} + R_{12}z + R_{12}^T z^* + R_{22}zz^* < 0\}$$

$$z = \operatorname{Re}(z) + j \operatorname{Im}(z), \quad z^* = \operatorname{Re}(z) - j \operatorname{Im}(z)$$

Circle with upper bound
of spectral radii \bar{r}

$$R_{11} = -\bar{r}^2$$

$$R_{12} = 0$$

$$R_{22} = 1$$

$$\implies \operatorname{Re}^2 + \operatorname{Im}^2 < \bar{r}^2$$

Angle-Ellipse with upper bound of damping angle $\bar{\vartheta}$

$$R_{11} = \begin{pmatrix} -1 & -\lambda_M/a & 0 & 0 \\ -\lambda_M/a & -1 & 0 & 0 \\ 0 & 0 & -2 \sin \bar{\vartheta} & 0 \\ 0 & 0 & 0 & -2 \sin \bar{\vartheta} \end{pmatrix}$$

$$R_{12} = \begin{pmatrix} 0 & (1/a - 1/b)/2 & 0 & 0 \\ (1/a + 1/b)/2 & 0 & 0 & 0 \\ 0 & 0 & \sin \bar{\vartheta} & \cos \bar{\vartheta} \\ 0 & 0 & -\cos \bar{\vartheta} & \sin \bar{\vartheta} \end{pmatrix}$$

$$R_{22} = 0$$

$$R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\textcolor{red}{P} \tilde{\mathcal{A}}_i)) + R_{22} \otimes (\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i) \prec 0$$

$$R_{11} \otimes P + \text{He}(R_{12} \otimes (P \tilde{\mathcal{A}}_i)) + R_{22} \otimes (\tilde{\mathcal{A}}_i^T P \tilde{\mathcal{A}}_i) \prec 0$$



Peaucelle et. al (2000)

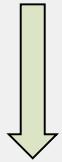
$$\begin{pmatrix} R_{11} \otimes P + \text{He}(F(I \otimes \tilde{\mathcal{A}}_i)) & R_{12} \otimes P + (I \otimes \tilde{\mathcal{A}}_i^T) G - F \\ \star & R_{22} \otimes P - G - G^T \end{pmatrix} \prec 0$$

$$R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\textcolor{red}{P} \tilde{\mathcal{A}}_i)) + R_{22} \otimes (\tilde{\mathcal{A}}_i^T \textcolor{red}{P} \tilde{\mathcal{A}}_i) \prec 0$$



Peaucelle et. al (2000)

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(\textcolor{red}{F}(I \otimes \tilde{\mathcal{A}}_i)) & R_{12} \otimes \textcolor{red}{P} + (I \otimes \tilde{\mathcal{A}}_i^T) \textcolor{red}{G} - \textcolor{red}{F} \\ \star & R_{22} \otimes \textcolor{red}{P} - \textcolor{red}{G} - \textcolor{red}{G}^T \end{pmatrix} \prec 0$$



Lerch et. al (2022)

$F = R_{12} \otimes \hat{P}$ and $G = R_{22} \otimes \hat{P}$ with $\hat{P} = \hat{P}^T = \text{const}$

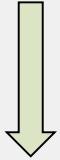
$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{22} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ \star & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} \prec 0$$

$$R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\textcolor{red}{P} \tilde{\mathcal{A}}_i)) + R_{22} \otimes (\tilde{\mathcal{A}}_i^T \textcolor{red}{P} \tilde{\mathcal{A}}_i) \prec 0$$



Peaucelle et. al (2000)

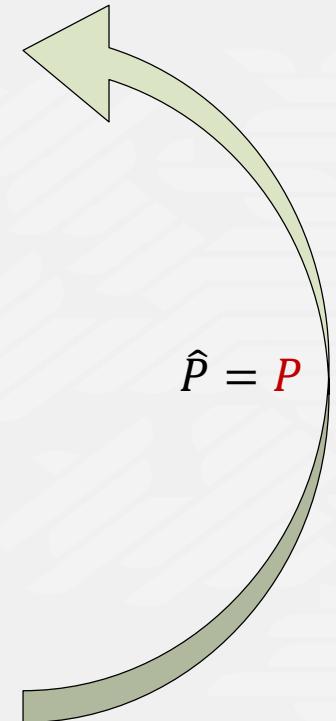
$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(\textcolor{red}{F}(I \otimes \tilde{\mathcal{A}}_i)) & R_{12} \otimes \textcolor{red}{P} + (I \otimes \tilde{\mathcal{A}}_i^T) \textcolor{red}{G} - \textcolor{red}{F} \\ * & R_{22} \otimes \textcolor{red}{P} - \textcolor{red}{G} - \textcolor{red}{G}^T \end{pmatrix} \prec 0$$



Lerch et. al (2022)

$F = R_{12} \otimes \hat{P}$ and $G = R_{22} \otimes \hat{P}$ with $\hat{P} = \hat{P}^T = \text{const}$

$$\boxed{\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{22} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} \prec 0}$$



$$\hat{P} = \textcolor{red}{P}$$

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ \star & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} \prec 0$$

Circle with radius \bar{r}

$$\begin{pmatrix} -\bar{r}^2 \textcolor{red}{P} & (\hat{P} \tilde{\mathcal{A}}_i)^T \\ \star & \textcolor{red}{P} - 2\hat{P} \end{pmatrix} \prec 0$$

$$\begin{aligned} R_{11} &= -\bar{r}^2 \\ R_{12} &= 0 \\ R_{22} &= 1 \end{aligned}$$

$$\Rightarrow \begin{pmatrix} 0 & \hat{P}^{-1} \\ I & 0 \end{pmatrix} \begin{pmatrix} -\bar{r}^2 \textcolor{red}{P} & (\hat{P} \tilde{\mathcal{A}}_i)^T \\ \star & \textcolor{red}{P} - 2\hat{P} \end{pmatrix} \begin{pmatrix} 0 & I \\ \hat{P}^{-1} & 0 \end{pmatrix} \prec 0$$

$$\Rightarrow \begin{pmatrix} \hat{P}^{-1}(2I - \textcolor{red}{P}\hat{P}^{-1}) & \tilde{\mathcal{A}}_i \\ \star & \bar{r}^2 \textcolor{red}{P} \end{pmatrix} \succ 0 \quad \equiv \text{Linearization Method}$$

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} < 0$$

$\textcolor{red}{P}$ is decoupled from $\tilde{\mathcal{A}}_i$: all considered controller types are possible

Optimum: $\hat{P} = P$

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} < 0$$

$\textcolor{red}{P}$ is decoupled from $\tilde{\mathcal{A}}_i$: all considered controller types are possible

Optimum: $\hat{P} = P$

Problem

Solution

P is not known beforehand

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} < 0$$

$\textcolor{red}{P}$ is decoupled from $\tilde{\mathcal{A}}_i$: all considered controller types are possible

Optimum: $\hat{P} = P$

Problem

P is not known beforehand

Solution

Initialization with $\hat{P} = I$
Iteration with $\hat{P}_{l+1} = P_l$

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} < 0$$

$\textcolor{red}{P}$ is decoupled from $\tilde{\mathcal{A}}_i$: all considered controller types are possible

Optimum: $\hat{P} = P$

Problem

P is not known beforehand

$\hat{P} = I$ might lead to infeasibility

Solution

Initialization with $\hat{P} = I$
Iteration with $\hat{P}_{l+1} = P_l$

$$\begin{pmatrix} R_{11} \otimes \textcolor{red}{P} + \text{He}(R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)) & R_{12} \otimes (\textcolor{red}{P} - \hat{P}) + R_{12} \otimes (\hat{P} \tilde{\mathcal{A}}_i)^T \\ * & R_{22} \otimes (\textcolor{red}{P} - 2\hat{P}) \end{pmatrix} < 0$$

$\textcolor{red}{P}$ is decoupled from $\tilde{\mathcal{A}}_i$: all considered controller types are possible

Optimum: $\hat{P} = P$

Problem

P is not known beforehand

$\hat{P} = I$ might lead to infeasibility

Solution

Initialization with $\hat{P} = I$
 Iteration with $\hat{P}_{l+1} = P_l$

Initialization with $\bar{r} > 1$ (unstable)
 Iterative minimization of \bar{r} until $\bar{r} = 1$ (stable)

Stage 1 Stabilize the closed loop

- Start with $\hat{P} = I$ and $\bar{r} > 1$
- End when $\bar{r} = 1$

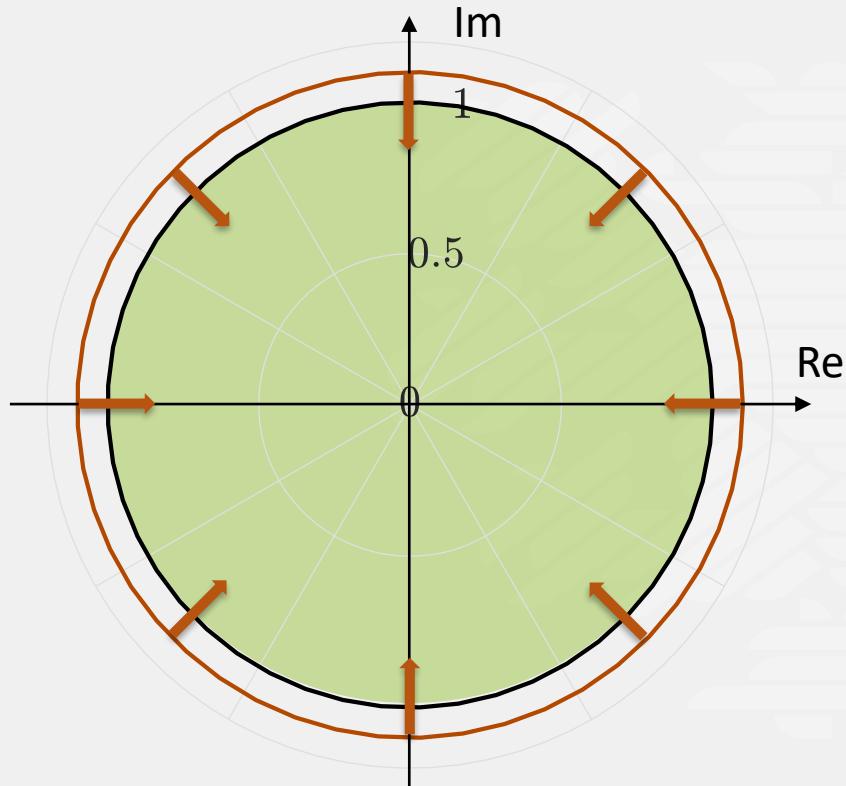
Stage 2 Minimize \bar{r}

Stage 3 Minimize ϑ

$$R_{11} = -\bar{r}^2$$

$$R_{12} = 0$$

$$R_{22} = 1$$



Stage 1 Stabilize the closed loop

Stage 2 Minimize \bar{r}

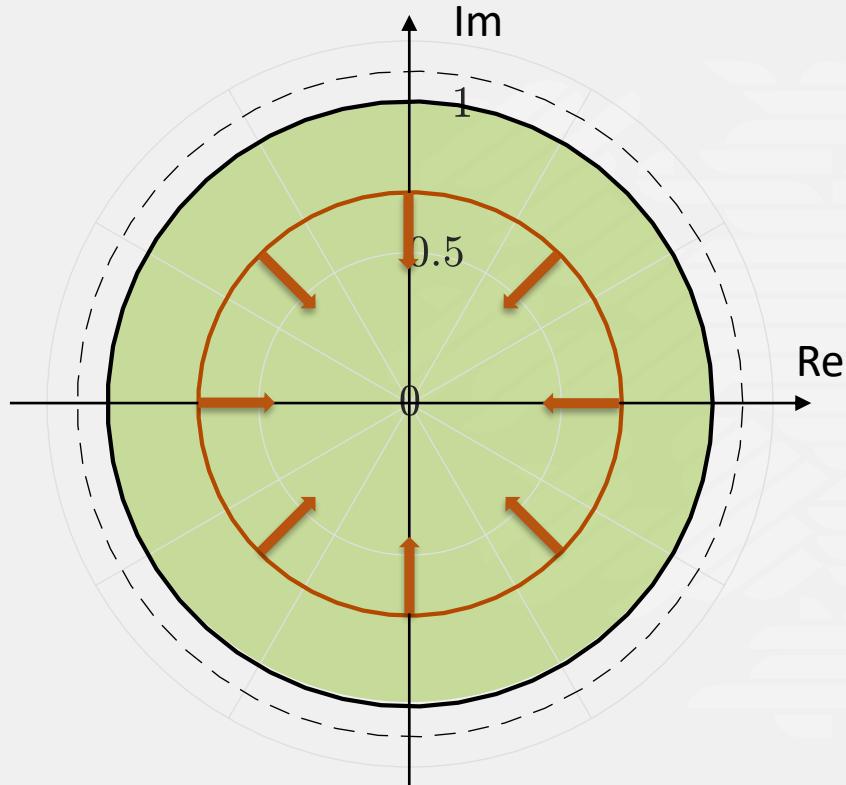
- Start with $\bar{r} = 1$
- End when \bar{r} cannot be reduced further

Stage 3 Minimize $\bar{\vartheta}$

$$R_{11} = -\bar{r}^2$$

$$R_{12} = 0$$

$$R_{22} = 1$$



Stage 1 Stabilize the closed loop

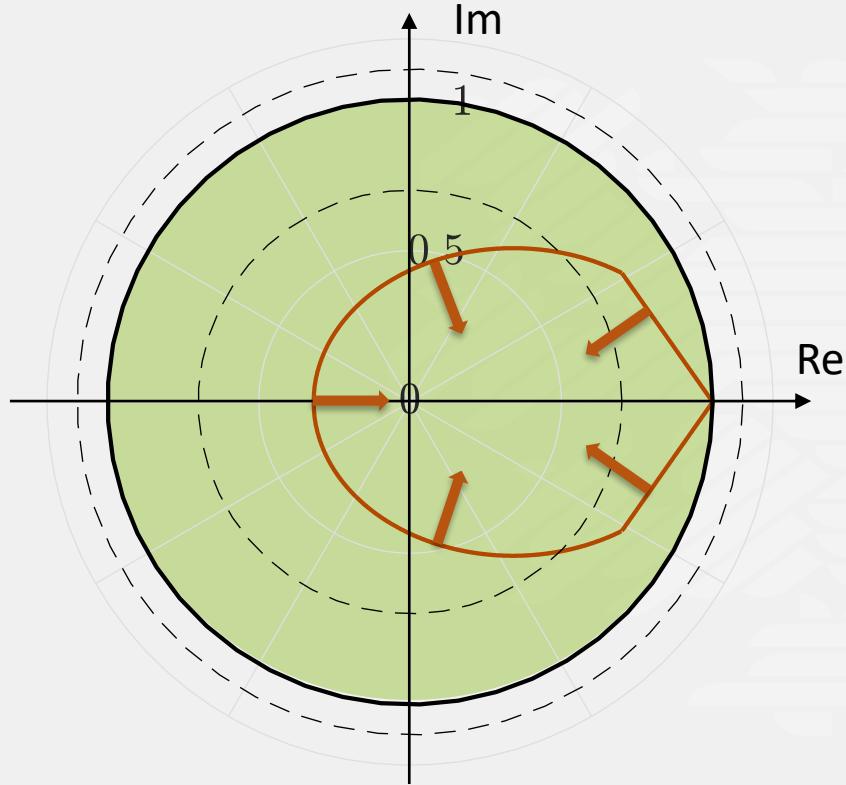
Stage 2 Minimize r

Stage 3 Minimize $\bar{\vartheta}$

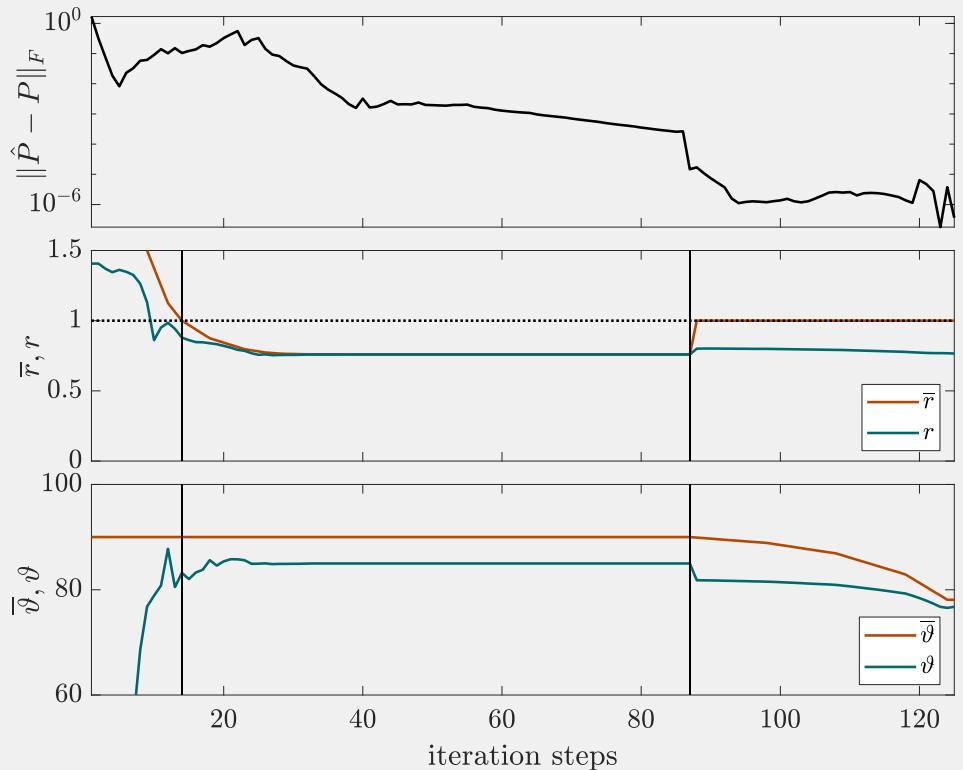
- Start with $\bar{\vartheta} = 90^\circ$
- End when $\bar{\vartheta}$ cannot be reduced further

$$R_{11} = \begin{pmatrix} -1 & -\lambda_M/a & 0 & 0 \\ -\lambda_M/a & -1 & 0 & 0 \\ 0 & 0 & -2 \sin \bar{\vartheta} & 0 \\ 0 & 0 & 0 & -2 \sin \bar{\vartheta} \end{pmatrix}$$

$$R_{12} = \begin{pmatrix} 0 & (1/a - 1/b)/2 & 0 & 0 \\ (1/a + 1/b)/2 & 0 & 0 & 0 \\ 0 & 0 & \sin \bar{\vartheta} & \cos \bar{\vartheta} \\ 0 & 0 & -\cos \bar{\vartheta} & \sin \bar{\vartheta} \end{pmatrix} \quad R_{22} = 0$$



Example 1: Optimization Procedure



Numerical example (Benzaouia (2006))

$$A_s = \begin{pmatrix} 1 & 1 \\ 0 & 1 \end{pmatrix}$$

$$x_0 = (1 \quad 1)^T$$

$$B_s = \begin{pmatrix} 0.5 \\ 1 \end{pmatrix}$$

$$u_{\max} = 1$$

$$C_s = (1 \quad 1)$$

$$v_{\max} = 1$$

Design of observer-based state feedback

$$K = (-0.0489 \quad -0.3358)$$

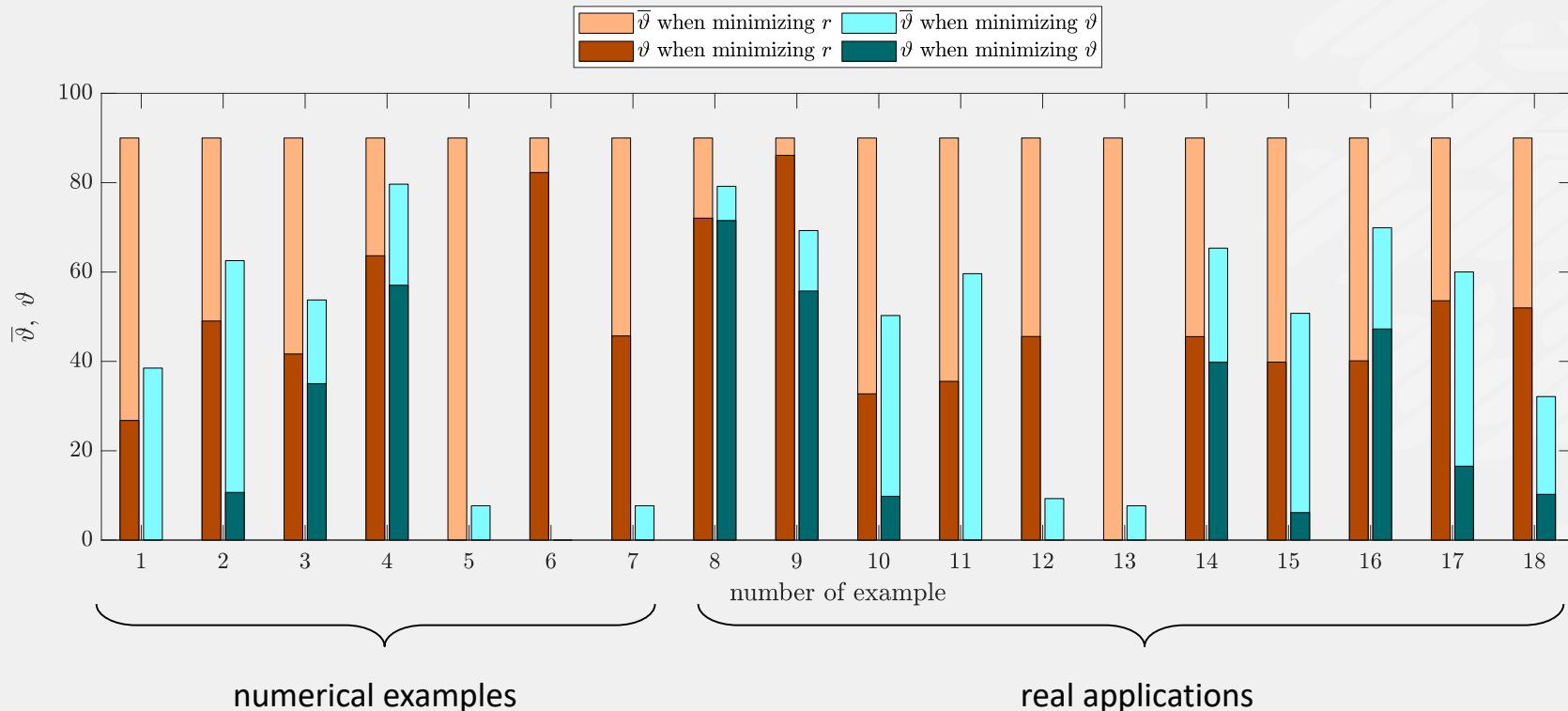
$$L = \begin{pmatrix} 0.6441 \\ 0.7594 \end{pmatrix}$$

$$r = 0.7678 \quad \vartheta = 76.5533^\circ$$

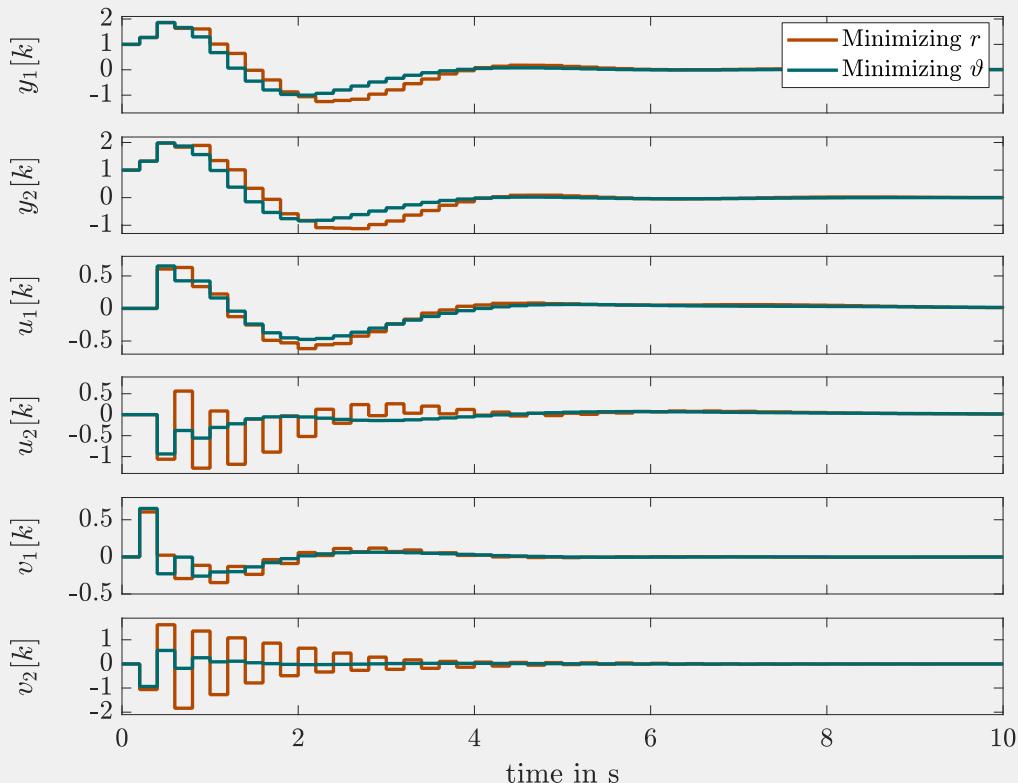
$$\|\hat{P} - P\|_F = 3.6335 \cdot 10^{-7}$$

18 Systems

Design of full state feedback



Example 3: Unstable Aircraft (Safonov et al. (1981))



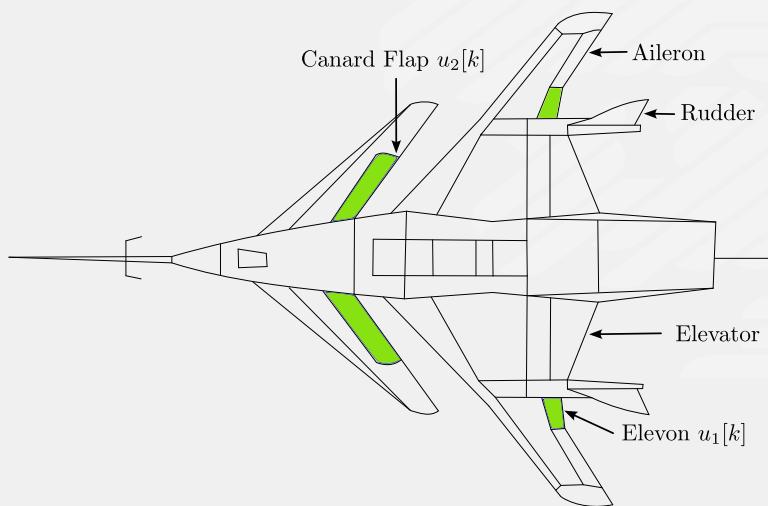
Design of observer-based state feedback

$$r = 0.9021$$

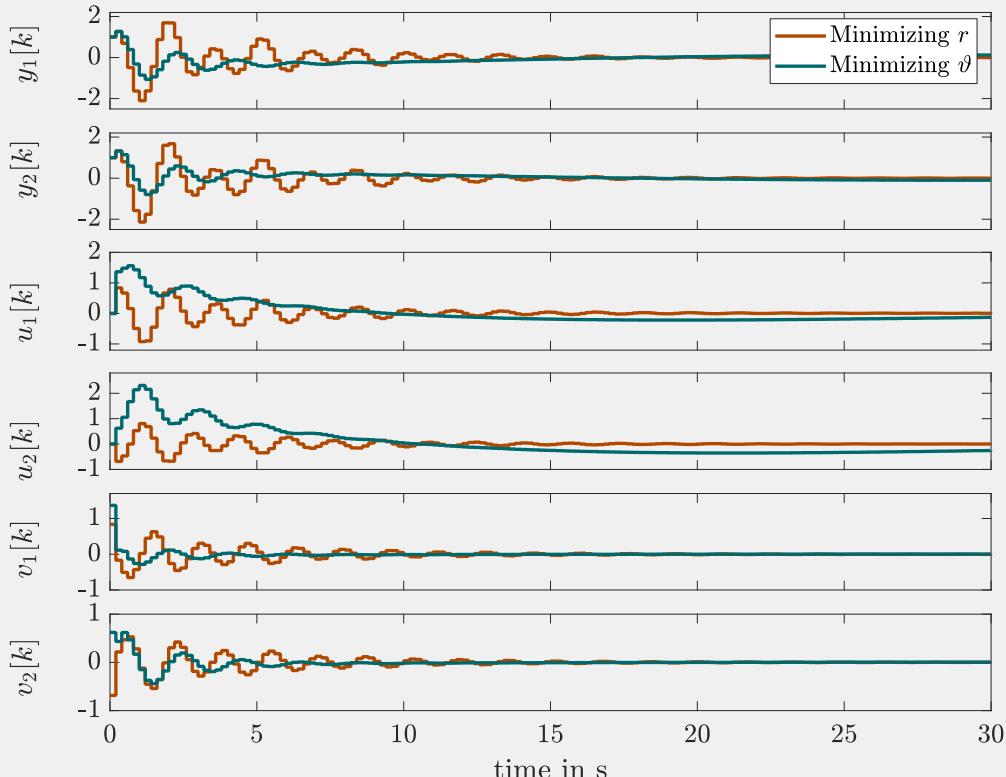
$$\vartheta = 86.8503^\circ$$

$$r = 0.9023$$

$$\vartheta = 70.6795^\circ$$



Example 4: Unstable Aircraft (Safonov et al. (1981))



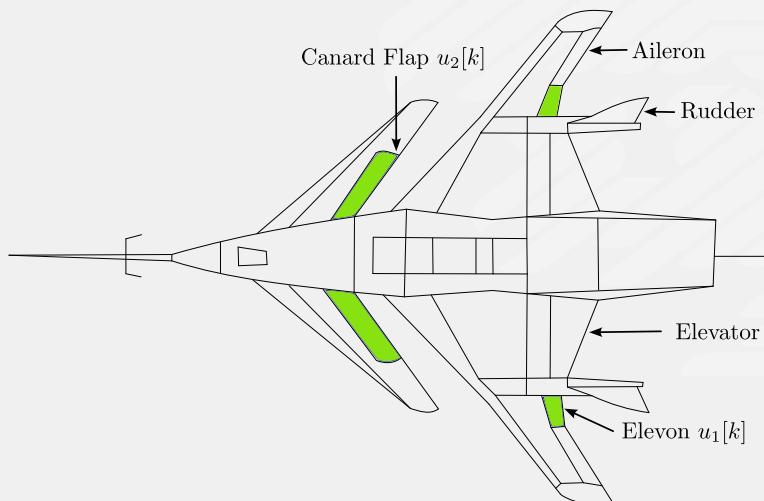
Design of PID controller

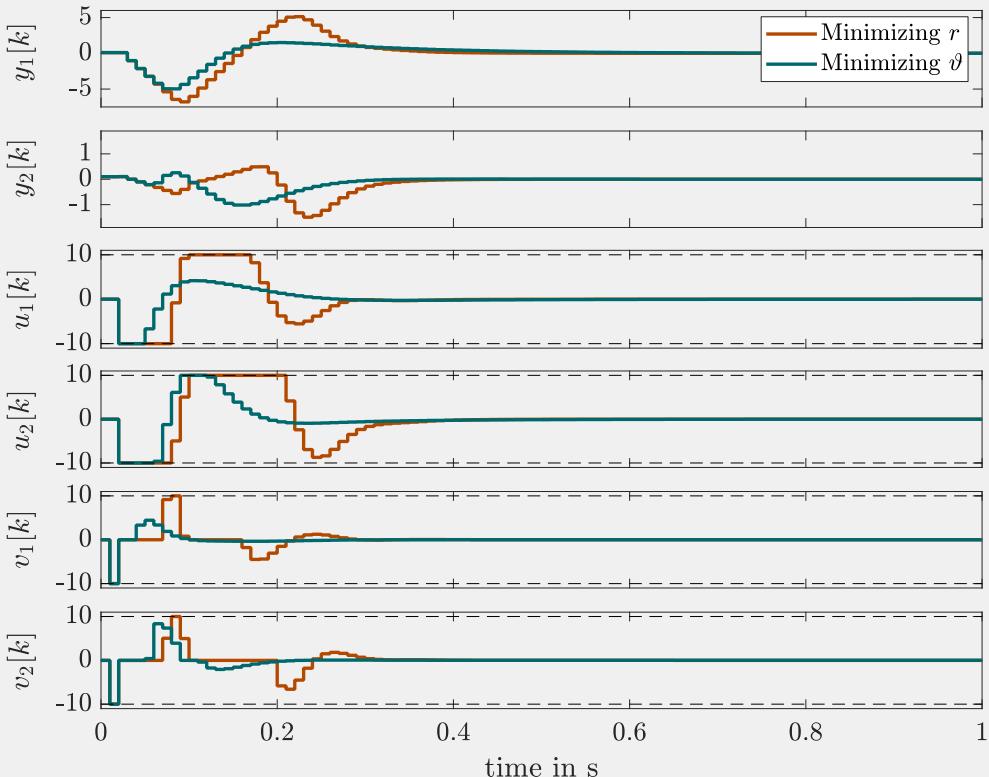
$$r = 0.9952$$

$$\vartheta = 87.0185^\circ$$

$$r = 0.9854$$

$$\vartheta = 81.2153^\circ$$





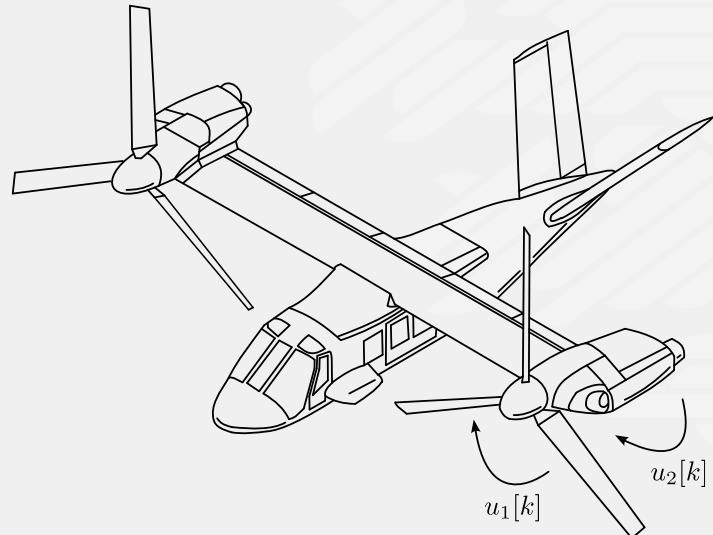
Design of observer-based state feedback

$$r = 0.7871$$

$$\vartheta = 70.2408^\circ$$

$$r = 0.9308$$

$$\vartheta = 27.1399^\circ$$



Summary

- Minimization of oscillations (D_R region pole placement)
- Magnitude and rate saturation
- Iterative LMI method
- Design of different controller types with one method

Outlook

- Other D_R regions for other objectives or combinations
- More accurate convex approximation of cardioid
- Tests on real applications
- Uncertainties, quasilinear forms, ...

Thank you for your attention!
Any questions?